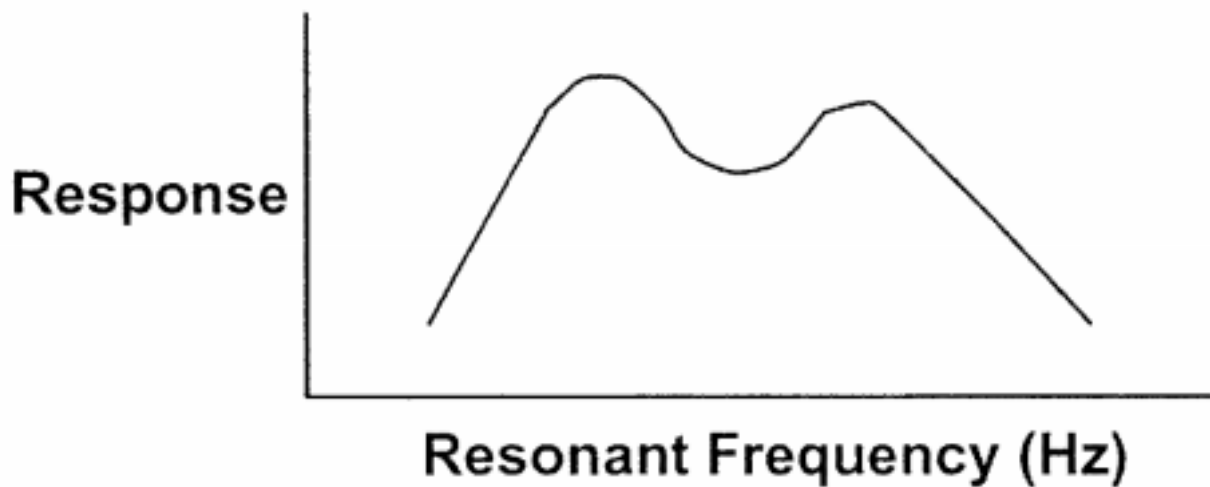


---

**WORKSHOP PROBLEM 9b**

***Response Spectra (cont.)***



**Objectives:**

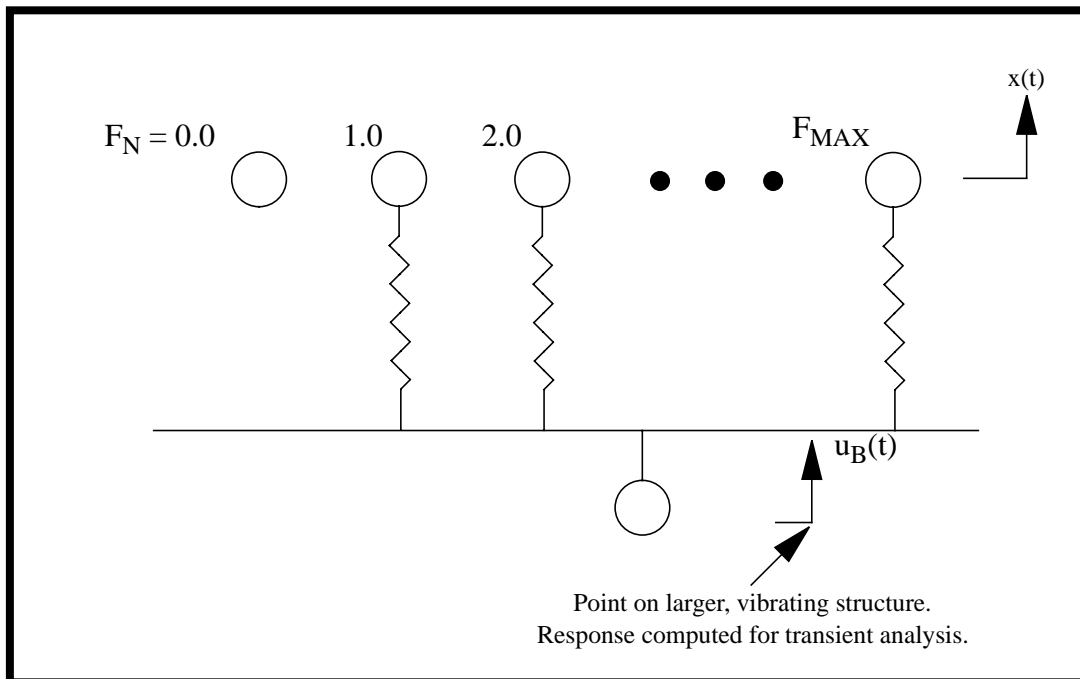
- Apply the shock spectrum.
- Submit the file for analysis in MSC/NASTRAN.
- Calculate the shock response using SOL 103.



**Model Description:**

Define the shock response of the plate due to a  $2.0 \text{ in/sec}^2$  sine pulse applied at the clamped edge. Use modes to a frequency of 1000 Hz with 3% critical damping. Use the SRSS option for model response summation.

**Figure 9b.1-Model Description and Loading Diagram**



---

## Suggested Exercise Steps:

- Reference a previously created dynamic math model, **plate.bdf**, by using the INCLUDE statement.
- Modify boundary conditions for clamped modes.
- Place big foundation mass (BFM) at base to simulate 'clamped' modes (CMASS2).
- RBE mass to remaining base point (RBE2).
- Identify excitation DOFs (SUPPORT).
- Specify damping table (TABDMP1).
- Specify shock spectrum to be used (DLOAD).
- Specify shock tables (DTI).
- Insert punch output for shock spectrum calculation.
- Specify the appropriate parameters.
  - PARAM, SCRSPEC, 0
  - PARAM, OPTION, SRSS
  - PARAM, LFREQ, 0.1
  - PARAM, HFREQ, 1000
- Generate an input file and submit it to the MSC/NASTRAN solver (SOL103).
- Review the results.

**ID SEMINAR,PROB9B**

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**CEND**

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1	2	3	4	5	6	7	8	9	10

**ENDDATA**

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# Generating an input file for MSC/NASTRAN Users:

MSC/NASTRAN users can generate an input file using the data from page 9-3 (Model Description). The result should be similar to the output below.

## 1. MSC/NASTRAN input file: **prob9b.dat**.

```
ID SEMINAR, PROB9B
SOL 103
TIME 30
CEND
TITLE= RESPONSE SPECTRUM ANALYSIS
SUBTITLE= USING CALCULATED SHOCK RESPONSE
LABEL= SHOCK WILL BE INPUT IN Z DIRECTION
ECHO= UNSORTED
SET 111= ALL
DISPLACEMENT= 111
SPC= 200
SUBCASE 1
METHOD= 100
SDAMP= 200
DLOAD= 500
$
BEGIN BULK
$
$ PLATE MODEL DESCRIBED IN NORMAL MODES EXAMPLE
$
INCLUDE 'plate.bdf'
PARAM,COUPMASS,1
PARAM,WTMASS,0.00259
$
$ BOUNDARY CONDITIONS FOR `CLAMPED' MODES
$
SPC1, 200, 1245, 1, 12, 23, 34, 45
$
$ PLACE BIG FOUNDATION MASS (BFM) AT BASE
$ TO STIMULATE `CLAMPED' MODES
$
CMASS2, 110, 1000., 23, 3
$
$ RBE MASS TO REMAINING BASE POINTS
$
RBE2, 101, 23, 3, 1, 12, 34, 45
```



```
$
$ SUPPORT CARD TO IDENTIFY EXCITATION DOFS
$
SUPPORT, 23, 3
$
$ EIGENVALUE EXTRACTION
$ MUST BE MASS NORMALIZED (DEFAULT)
$
EIGR, 100, MGIV, 0., 1000.
$
$ TABLE TO SPECIFY DAMPING FOR USE IN THE ANALYSIS
$
TABDMP1, 200, CRIT,
, 0., 0.03, 1000., 0.03, ENDT
$
$ SPECIFICATION OF SHOCK SPECTRUM TO BE USED
$
DLOAD, 500, 1.0, 2.0, 1
$
$ DLOAD, ID, OVERALL SCALE, SCALE FOR R-SET DOF# 1, SHOCK TABLE FOR DOF# 1,
$ SCALE FOR R-SET DOF# 2, SHOCK TABLE FOR DOF# 2, ETC.
$
$ SELECT SHOCK RESPONSE CALCULATION
$
PARAM, SCRSPEC, 0
$
$ SELECT SUMMATION OPTION
$
PARAM, OPTION, SRSS
$
$ MODAL FREQUENCY RANGE CAN BE SELECTED USING
PARAM, LFREQ, 0.1
PARAM, HFREQ, 1000.
$
$ SPECIFICATION FOR SHOCK TABLES
$
DTI, SPECSEL, 0
DTI, SPECSEL, 1, , A, 2, 0., 3, 0.02,
, 4, 0.04, ENDREC
$
$ DTI, SPECSEL, SHOCK TABLE NUMBER, , [(A)CCELERATION, (V)ELOCITY, OR (D)ISP],
$ TABLED1 POINTER, DAMPING FOR TABLE, ETC.
$
```

---

\$ PUNCH OUTPUT FOR SHOCK SPECTRUM CALCULATION

\$

\$ ACCE 4 3000 3 1

\$ 0.000000E+00

\$

TABLED1 2

20.	.038683	40.	.152539	60.	.33511	80.	.576059
100.	.862049	120.	1.17619	140.	1.50169	160.	1.82018
180.	2.11404	200.	2.36801	220.	2.56617	240.	2.70027
260.	2.76275	280.	2.75073	300.	2.74632	320.	2.61887
340.	2.4218	360.	2.39068	380.	2.24931	400.	2.02296
420.	1.78538	440.	1.70355	460.	1.57056	480.	1.40493
500.	1.22608	520.	1.20483	540.	1.17631	560.	1.14097
580.	1.10048	600.	1.05582	620.	1.00818	640.	.958761
660.	.908725	680.	.859158	700.	.827667	720.	.782127
740.	.728996	760.	.694088	780.	.668602	800.	.635044
820.	.598496	840.	.571831	860.	.563072	880.	.550499
900.	.528854	920.	.509281	940.	.500534	960.	.498016
980.	.488793	1000.	.468321	ENDT			

\$ACCE 4 3000 3 52

\$ 2.000000E-02

TABLED1 3

20.	.037708	40.	.143365	60.	.314936	80.	.541342
100.	.80976	120.	1.10506	140.	1.40671	160.	1.69567
180.	1.98167	200.	2.22217	220.	2.35249	240.	2.53055
260.	2.56231	280.	2.55577	300.	2.58668	320.	2.45921
340.	2.29411	360.	2.25956	380.	2.12901	400.	1.92605
420.	1.68656	440.	1.61355	460.	1.4968	480.	1.35263
500.	1.19796	520.	1.17707	540.	1.14947	560.	1.11613
580.	1.07807	600.	1.03637	620.	.992124	640.	.946383
660.	.900171	680.	.854434	700.	.810016	720.	.767647
740.	.727923	760.	.691288	780.	.658039	800.	.628311
820.	.602091	840.	.579207	860.	.559362	880.	.542128
900.	.526973	920.	.51329	940.	.500403	960.	.487602
980.	.474171	1000.	.459408	ENDT			

\$ACCE 4 3000 3 103

\$ 4.000000E-02

TABLED1 4

20.	.039336	40.	.137673	60.	.297382	80.	.511244
100.	.764891	120.	1.04406	140.	1.31588	160.	1.58461
180.	1.85678	200.	2.10175	220.	2.19165	240.	2.3921
260.	2.39929	280.	2.42782	300.	2.44263	320.	2.317
340.	2.17923	360.	2.14283	380.	2.0227	400.	1.8407
420.	1.62279	440.	1.53417	460.	1.43168	480.	1.30597
500.	1.17212	520.	1.15165	540.	1.12513	560.	1.09349

580.	1.05768	600.	1.01868	620.	.977462	640.	.934986
660.	.892143	680.	.849752	700.	.808538	720.	.769114
740.	.731968	760.	.69746	780.	.665814	800.	.637115
820.	.611319	840.	.588261	860.	.567655	880.	.549125
900.	.532205	920.	.516369	940.	.501047	960.	.485644
980.	.469568	1000.	.452243	ENDT			

\$

ENDDATA

---

## Submitting the input file for analysis:

2. Submit the input file to MSC/NASTRAN for analysis.

To submit the MSC/NASTRAN **.dat** file, find an available UNIX shell window and at the command prompt enter **nastran prob9b scr=yes**. Monitor the run using the UNIX **ps** command.

3. When the run is completed, edit the **prob9b.f06** file and search for the word **FATAL**. If no matches exist, search for the word **WARNING**. Determine whether existing **WARNING** messages indicate modeling errors.

## Comparison of Results

4. Compare the results obtained in the **.f06** file with the results on the following page:

\*\*\* USER INFORMATION MESSAGE 3035 FOR DATA BLOCK KLR

SUPPORT PT.NO. EPSILON STRAIN ENERGY EPSILONS LARGER THAN 0.001 ARE FLAGGED WITH ASTERISKS  
 1 9.5369362E-15 2.2118911E-09  
 RESPONSE SPECTRUM ANALYSIS JULY 5, 1997 MSC/NASTRAN 1/23/97 PAGE 12  
 USING CALCULATED SHOCK RESPONSE  
 SHOCK WILL BE INPUT IN Z DIRECTION SUBCASE 1

\*\*\* USER INFORMATION MESSAGE 4415,  
 THE FOLLOWING A-SET DEGREES OF FREEDOM HAVE EITHER NULL MASSES OR NULL MASSES AND STIFFNESS.  
 RESPONSE SPECTRUM ANALYSIS JULY 5, 1997 MSC/NASTRAN 1/23/97 PAGE 13  
 USING CALCULATED SHOCK RESPONSE  
 SHOCK WILL BE INPUT IN Z DIRECTION SUBCASE 1

MODE NO.	EXTRACTION ORDER	EIGENVALUE	R E A L E I G E N V A L U E S		GENERALIZED	
			RADIANS	CYCLES	MASS	STIFFNESS
1	101	0.0	0.0	0.0	1.000000E+00	0.0
2	102	7.058213E+05	8.401317E+02	1.337111E+02	1.000000E+00	7.058213E+05
3	103	1.878432E+07	4.334088E+03	6.897916E+02	1.000000E+00	1.878432E+07
4	104	2.811620E+07	5.302471E+03	8.439145E+02	1.000000E+00	2.811620E+07
5	105	1.931709E+08	1.389859E+04	2.212030E+03	0.0	0.0
6	1	2.234434E+08	1.494802E+04	2.379052E+03	0.0	0.0
7	106	2.328846E+08	1.526056E+04	2.428793E+03	0.0	0.0
8	107	6.845925E+08	2.616472E+04	4.164244E+03	0.0	0.0
9	108	9.602310E+08	3.098759E+04	4.931829E+03	0.0	0.0
10	109	1.365612E+09	3.695418E+04	5.881441E+03	0.0	0.0

^^^ USER INFORMATION MESSAGE 9047 (POSTREIG) - SCALED RESPONSE SPECTRA FOR RESIDUAL STRUCTURE ONLY  
 RESPONSE SPECTRUM ANALYSIS JULY 5, 1997 MSC/NASTRAN 1/23/97 PAGE 20  
 USING CALCULATED SHOCK RESPONSE  
 SHOCK WILL BE INPUT IN Z DIRECTION SUBCASE 1

MATRIX FN (GINO NAME 101 ) IS A DB PREC 1 COLUMN X 3 ROW RECTANG MATRIX.  
 COLUMN 1 ROWS 1 THRU 3 -----  
 ROW

1) 1.337111500777D+02 6.897918198043D+02 8.439147827213D+02  
 THE NUMBER OF NON-ZERO TERMS IN THE DENSEST COLUMN = 3  
 THE DENSITY OF THIS MATRIX IS 100.00 PERCENT.

RESPONSE SPECTRUM ANALYSIS JULY 5, 1997 MSC/NASTRAN 1/23/97 PAGE  
 USING CALCULATED SHOCK RESPONSE  
 SHOCK WILL BE INPUT IN Z DIRECTION SUBCAS

PSIT  
 POINT VALUE POINT VALUE POINT VALUE POINT VALUE POINT VALU

COLUMN 1  
 23 T3 -2.11560E-02  
 COLUMN 2  
 23 T3 -4.40398E-16  
 COLUMN 3  
 23 T3 1.18599E-02

RESPONSE SPECTRUM ANALYSIS JULY 5, 1997 MSC/NASTRAN 1/23/97 PAGE  
 USING CALCULATED SHOCK RESPONSE  
 SHOCK WILL BE INPUT IN Z DIRECTION SUBCAS

U S E T D E F I N I T I O N T A B L E ( I N T E R N A L S E Q U E N C E , R O W S O R  
 R D I S P L A C E M E N T S E T  
 -1- -2- -3- -4- -5- -6- -7- -8- -9- -

1= 23-3  
 SCALED SPECTRAL RESPONSE, SRSS OPTION, DLOAD = 500 CLOSE = 1.00

RESPONSE SPECTRUM ANALYSIS JULY 5, 1997 MSC/NASTRAN 1/23/97 PAGE  
 USING CALCULATED SHOCK RESPONSE  
 SHOCK WILL BE INPUT IN Z DIRECTION SUBCAS

MATRIX UHVR (GINO NAME 101 ) IS A REAL 3 COLUMN X 3 ROW SQUARE MATRIX.

COLUMN 1 ROWS 1 THRU 3 -----  
 ROW  
 1) 7.6201E-08 0.0000E+00 4.8912E-10  
 COLUMN 2 ROWS 1 THRU 3 -----  
 ROW  
 1) 6.4019E-05 0.0000E+00 2.5935E-06  
 COLUMN 3 ROWS 1 THRU 3 -----  
 ROW  
 1) 5.3784E-02 0.0000E+00 1.3752E-02

THE NUMBER OF NON-ZERO TERMS IN THE DENSEST COLUMN = 2  
 THE DENSITY OF THIS MATRIX IS 66.67 PERCENT.

RESPONSE SPECTRUM ANALYSIS JULY 5, 1997 MSC/NASTRAN 1/23/97 PAGE

MSC/NASTRAN 102 Exercise Workbook

USING CALCULATED SHOCK RESPONSE  
 SHOCK WILL BE INPUT IN Z DIRECTION  
 TIME = 0.000000E+00

SUBCASE 1

## D I S P L A C E M E N T V E C T O R

POINT ID.	TYPE	T1	T2	T3	R1	R2	R3
1	G	0.0	0.0	6.222643E-10	0.0	0.0	0.0
2	G	0.0	0.0	7.374101E-08	8.308994E-08	3.059527E-07	0.0
3	G	0.0	0.0	3.173752E-07	1.214202E-07	6.616859E-07	0.0
4	G	0.0	0.0	7.194814E-07	1.167081E-07	9.376687E-07	0.0
5	G	0.0	0.0	1.246589E-06	1.039592E-07	1.160147E-06	0.0
6	G	0.0	0.0	1.870825E-06	8.258849E-08	1.326412E-06	0.0
7	G	0.0	0.0	2.565942E-06	6.149954E-08	1.444516E-06	0.0
8	G	0.0	0.0	3.308983E-06	4.203442E-08	1.519611E-06	0.0
9	G	0.0	0.0	4.080410E-06	2.671514E-08	1.559985E-06	0.0
10	G	0.0	0.0	4.865109E-06	1.691765E-08	1.575028E-06	0.0
11	G	0.0	0.0	5.653596E-06	1.301237E-08	1.577945E-06	0.0
12	G	0.0	0.0	6.222643E-10	0.0	0.0	0.0
13	G	0.0	0.0	9.569825E-08	1.430303E-08	3.703413E-07	0.0
14	G	0.0	0.0	3.577567E-07	4.357941E-08	6.684602E-07	0.0
15	G	0.0	0.0	7.614155E-07	5.321706E-08	9.362424E-07	0.0
16	G	0.0	0.0	1.284521E-06	4.871796E-08	1.147647E-06	0.0
17	G	0.0	0.0	1.901587E-06	4.101944E-08	1.312728E-06	0.0
18	G	0.0	0.0	2.589113E-06	3.141624E-08	1.430590E-06	0.0
19	G	0.0	0.0	3.325147E-06	2.263403E-08	1.507938E-06	0.0
20	G	0.0	0.0	4.091029E-06	1.555984E-08	1.551409E-06	0.0
21	G	0.0	0.0	4.872190E-06	1.101371E-08	1.570698E-06	0.0
22	G	0.0	0.0	5.659176E-06	8.953050E-09	1.576591E-06	0.0
23	G	0.0	0.0	6.222643E-10	0.0	0.0	0.0
24	G	0.0	0.0	9.883527E-08	0.0	3.798185E-07	0.0
25	G	0.0	0.0	3.680177E-07	0.0	6.836093E-07	0.0
26	G	0.0	0.0	7.745986E-07	0.0	9.334775E-07	0.0
27	G	0.0	0.0	1.296574E-06	0.0	1.145700E-06	0.0
28	G	0.0	0.0	1.911807E-06	0.0	1.307858E-06	0.0
29	G	0.0	0.0	2.596946E-06	0.0	1.426281E-06	0.0
30	G	0.0	0.0	3.330809E-06	0.0	1.503975E-06	0.0
.							
.							